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Damage characteristic of aluminum nanorod under hypervelocity im-

² pact

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Abstract

Understanding the dynamic behavior of materials under hypervelocity impact is of great impor-10 tance to develop new materials or structures for protective applications. The present work gives 11 insight into the damage characteristic of aluminum nanorod under hypervelocity impact based on 12 atomistic simulations. First of all, the propagation of impact wave is found to experience a rapid 13 decaying because of its release from the side surface, which leads to a complex three-dimensional 14 stress wave and two tension regions inside the nanorod. The damage mode under this tension state 15 is found to be very different from the classical spallation. Due to the interaction of two release 16 waves from the side and end surfaces, a temporary spall damage is observed and its initial tensile 17 strength is close to that of bulk material. However, that early spall damage does not develop into 18 a complete spall fracture. More importantly, all generated voids are found to be closed eventually 19 after their coalescence. Furthermore, the mass continues expanding outward from the impact plane 20 and finally causes a radial annular fragmentation. The annular fragmentation shows a clear crys-21 talline direction dependence for low impact velocities. The number and the size of final fragments 22 are found to follow a power law relationship for all impact velocities. 23

24 Keywords

²⁵ nanorod, hypervelocity impact, damage, atomistic simulation

²⁶ Introduction

Spallation is a typical failure case in studying the mechanisms those underlie and govern material 27 strength, and spallation is essentially attributed to the void nucleation, growth and coalescence for 28 ductile metals[1]. In the past decades, it has attracted abundant experimental, theoretical and com-29 putational researches. In experiments, plate impact[2-4] and laser shock [5-7] techniques are com-30 monly used to detect the spall damage with high strain rates ranging about from 10^5 to $10^{10}s^{-1}$. 31 Previous studies have shown that the spallation strength depends on not only the shock pulse[3] 32 and strain rate[6], but also the microstructure of material, such as crystal orientation[2] and grain 33 size[7]. However, the required spatiotemporal resolutions for a comprehensive understanding of the 34 void nucleation and growth is still prohibitive for state-of-the-art experimental techniques. In terms 35 of theoretical research, many empirical models have been developed to describe the spallation 36 based on experimental data and numerical simulations[8-10]. For instance, Czarnota et al.[10] re-37 produced the experimental results for different thickness flyers and different impact velocities very 38 well by using Weibull function to describe the stress distribution of void nucleation. Of course, 39 the dynamic damage in materials under extreme condition is often coupled with complex defor-40 mation, heat generation and even phase change. Thus, it is very hard to consider all the influence 41 factors at the same time in constructing a theoretical damage constitutive model, and there is still 42 no general spall criterion. Correspondingly, numerical simulation methods at different physical 43 scales have been developed to probe the damage dynamics of materials. Such as the finite element 44 methods[11,12] within the framework of the continuum mechanics, the recently developed quasi-45 coarse-grained dynamics [13,14] and molecular dynamics (MD) simulations. 46

At present, MD simulation has become a resourceful tool in studying various deformation mechanisms in materials at atomistic level. Recent MD researches[15-22] have shown many microstructure effects related to spallation, such as the grain size, anisotropy and intrinsic defects. For exam-

ple, it is found that when grain size decreases continuously to lower than 20 nm, the yield strength 50 of materials will change from an increasing trend to a decreasing trend, triggering the so-called in-51 verse Hall-Petch effect[7,23-25]. Remington et al.[7] found that the spallation in nanocrystalline 52 tantalum is primarily intergranular under relatively low strain rate deformation and hence the ef-53 fect of grain size is inverse Hall-Petch regulation. In practice, the dynamic response of nanoma-54 terials can exhibit more diverse characteristics because of the strong surface and crystal orienta-55 tion effects. The properties of the compressed surface are strongly dependent on a range of im-56 pact parameters, including impact velocity, material of incident particle, crystal orientation and 57 temperature, etc. For example, Lee et al. [26] conducted a microscopic ballistic test to study the 58 shock responses of nanocomposites and prominently show the orientation-dependence of energy 59 dissipation. Pogorelko et al.[27] investigated the impact process of copper nanoparticle with alu-60 minum surface and obtain the optimal range of temperature and collision velocity for producing 61 high-quality layers of copper on the aluminum surface. These effects are becoming more and more 62 concerned in relative application fields, such as the aerosol deposition and cold spraying[28], space 63 debris, etc. 64

Overall, the specific research on the dynamic mechanism of nanoscale materials is still in its in-65 fancy. In this regard, this work aims to exploit the dynamic damage characteristic in nanorod un-66 der planar impact loading. Here, lightweight face-centered cubic aluminum is selected, since alu-67 minum and aluminum hybrid matrix composites are promising candidates for a mass of structural 68 applications attributed to their excellent mechanical properties [29,30]. Based on the simulation re-69 sults, the decaying in stress propagation, void nucleation and evolution, as well as fragmentation 70 power law are all revealed. The structure of manuscript is as follows: Simulation details are intro-71 duced in Section 2, results and discussion are presented in Section 3, and finally summaries are 72 drawn in Section 4. 73

3

74 Simulation details

The empirical potential or force field is the physical basis of MD simulations, which determines the 75 accuracy and authenticity of results. Here, we choose the embedded atom method (EAM) potential 76 improved by Zhakhovskii et al.[31], which is suitable to simulate the behavior of aluminum crystal 77 under strong dynamic compression and tension. The initial configuration is a single crystal alu-78 minum nanorod with 1,360,329 atoms, whose diameter is 60a and length is 120a. The left 30a is 79 set as a flyer region and the last 90a as a target region, as shown in Fig.1. Here a is the lattice con-80 stant and a = 0.4032 nm. The x, y and z axes are along [100], [010] and [001] orientations, respec-81 tively. Our MD simulations are performed with the open-source LAMMPS[32] code. To obtain a 82 stable nanorod structure, the system is firstly relaxed by energy minimization with conjugate gradi-83 ent method, and then fully equilibrated in isothermal isobaric ensemble (NPT) with Nosé-Hoover 84 thermostat until it comes to minimum energy state. The relaxation temperature is set to 50 K to 85 reduce thermal noise to some extent. The impact processes are simulated in micro-canonical en-86 semble (NVE) and free boundaries are set in all the three directions. The impact velocity U_f varies 87 from 4 km/s to 7 km/s, which is added on the flyer region along the z direction. The time step in all 88 simulations is 1 fs. 89



Figure 1: Three-dimensional view of the single crystal aluminum nanorod. The diameter is 60a and length is 120a. The left 30a is set as the flyer region with impact velocity U_f and the other 90a is the target region. S_F and S_T denote the left and right end surfaces.

⁹⁰ Based on the atomic stress and microstructure analysis, we discuss the shock wave propagation,

tensile strength, damage evolution and fragmentation. The atomic stress is calculated according to virial formula[33]. And the shock pressure is obtained by averaging the stress components in the *x*, *y* and *z* directions, i.e., $P = -(\sigma_{xx} + \sigma_{yy} + \sigma_{zz})/3$. The atomic visualizations are performed by Ovito(the Open Visualization Tool)[34], and specifically, the void surface morphology is visualized with the construct surface mesh tool(Ovito)[35], which can also provide the value of surface areas. The number and size of clusters during fragmentation process are obtained with cluster analysis tool(Ovito).

Results and discussion

⁹⁹ The decaying property of impact wave

Above of all, we compare our simulations with the shock Hugoniot relations. The pressure peaks from our direct simulations are 40-98 GPa at the impact velocity 4-7 km/s, which agree well with the Hugoniot pressure calculated from $P_H = \rho_0 u_s u_p$. Here, ρ_0 is the initial sample density, u_s is the shock wave velocity and u_p is the particle velocity. These results indicate that the current impact scenario is almost an one-dimensional planar impact at the early stage.



Figure 2: Views of atomic stress distribution at different moments (a-d) after impact ($U_f = 7$ km/s). The release of shock wave at side surface is marked by white arcs (a), which changes the shocked stress distribution significantly. And interestingly, two different tension regions (b and d) can be observed inside the nanorod. The shock front is illustrated by gray lines.

¹⁰⁵ The time evolution of microscopic pressure distributions is presented in Fig.2 by considering the

case of $U_f = 7$ km/s. We firstly observe the formation of double shock waves after impact. Also, 106 the expansion of impact plane and the pressure release from the side surface already become dis-107 tinct at about 1.0 ps (Fig.2 (a)). At 2.0 ps (Fig.2 (b)), there appears the first tension region (Tension 108 I) near the left end surface, caused by the interaction between the two release waves from left end 109 and side surfaces. This early stage is similar to the typical spallation. The void nucleation inside 110 Tension I region occurs between 3.0 to 4.4 ps, as shown in Fig.2 (c) and (d). However, the planar 111 wave turns into a complex three-dimensional waves with time going. And we observe another ten-112 sion region (Tension II) in the target region. Apparently, the Tension II region is distinctly larger 113 than the Tension I region, and the shock pressure peak is already significantly reduced when it ar-114 rives at the right end surface. Thus, the damage state in target region is relatively weak. 115



Figure 3: Time evolution of pressure profile for (a) $U_f = 4$ km/s and (b) 7 km/s, respectively. P_z means the average pressure of each bin along z direction.

Based on the above dynamic process, we examine the pressure profiles for $U_f = 4$ km/s and 7 km/s by estimating the average pressure of each bin(~0.5 *a*) along z direction, and the results are shown in Fig.3. Due to the strong surface release effect, the peak pressure will experience an evident decaying. Our simulations show that the pressure peak changes from 40(98) GPa to 30(60) GPa for

 $U_f = 4(7)$ km/s within 1.0 ps. As illustrated in Fig.2, there forms a radial release wave propagates 120 inwards and meantime results in a dramatic expansion of the impact plane. Note that, the decaying 121 spontaneously renders a unique double peaks phenomenon. And the stress wave reflection from the 122 left end surface causes the Tension I in the flyer region. Moreover, with the propagation of differ-123 ent release waves, we observe the formation of Tension II, whose tension stress peak (~ 4 GPa) is 124 slightly larger than that in Tension I(\sim 3 GPa). Eventually, the peak pressure reduces to less than 125 10 GPa when the stress wave spreads to the right end surface in both cases of $U_f = 4$ km/s and 7 126 km/s. 127



Figure 4: Time evolution of velocity profile for (a) $U_f = 4$ km/s and (b) 7 km/s, respectively. The two microscopic views are sliced from center axis.

Further, we present in Fig.4 the axial velocity profiles at different times $U_f = 4$ km/s and 7 km/s, respectively. As described above, the release from the side surface will affect the impact wave profiles significantly. Here, we find that the particle velocity in shocked region seems to follow a linear distribution in the early stage, and its profile becomes irregular and form a complex threedimensional distribution with the impact wave propagation.

¹³³ Surface velocity history and tensile strength

Tensile strength is crucial to measure dynamic properties of materials. Spallation is a typical failure case in studying the mechanisms, which occurs where the two strong release waves meet. The entire spall process in ductile metal consists of the nucleation, growth, and coalescence of voids[1]. And the maximum of tensile stress during the spall process is regarded as the spall strength. In practice, the tensile strength is often deduced from the free surface velocity history based on acoustic approximation[1], since the *in situ* detection of spall stress is extremely difficult.



Figure 5: Time evolution of two end surface velocities. v_F (a and c) and v_T (b and d) represent the left and right end surface velocities along *z* direction respectively. The points A, B and C denote the velocity peak and valley at the early time.

In Fig.5, we present velocity history of the two end surfaces, i. e. the left (v_F) and right (v_T) surface velocities, by taking the cases of the $U_f = 4$ km/s and 7 km/s. Considering the severe deformation in the end surface during the impact process, the end surfaces are divided into three parts, including the inside part with $r \le 6$ nm, the middle part with $6 < r \le 10$ nm and the outside part with r > 10 nm. For both left and right end surfaces, the surface experiences a change from peak to valley. Nevertheless, there is remarkable difference between the two end surfaces during the following

process. For left end surface S_F , a remarkable pullback with a period of ~ 10 ps is observed, which 146 is followed by a velocity plateau (since the flyer region rapidly melts on the shock release and thus 147 there hardly form the stress wave reflection, as shown in Fig.5 (a) and Fig.5 (c)). For right end sur-148 face S_T , its velocity history can be characterized into three distinct stages (Fig.5 (b) and Fig.5 (d)), 149 i. e. oscillated acceleration, smooth acceleration and final velocity plateau, which reflect various 150 evolutions of inside damage. After being shocked, the velocity of S_T is found to decay from A to 151 B rapidly with the interactions of release waves from the side and end surfaces, and then rises to 152 C, resulting from that the new voids surfaces reflect the release wave into the compression wave. 153 The compression wave also leads to the voids coalescence at the target region. In other words, the 154 oscillation includes both the void nucleation and the wave reflection from the material end and side 155 surfaces. After all the voids disappear (at around 13.0 ps), this kind of oscillation decreases and the 156 acceleration becomes smooth. Finally, as the energy dissipation of waves and degree of material 157 breakage increase, the velocity of S_T increases gradually and reaches a velocity plateau. 158 The spall strength from surface and bulk are both analyzed here. As expected, the velocity in the 159

central region of the end surface experiences the most significant change, so we choose the inside part to estimate the tensile strength with the acoustic approximation[1] according to Eq.1. Meanwhile, the tensile strength from bulk is determined by Eq.2 with the maximum tensile stress during impact.

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 $\sigma_s = 0.5 \rho_0 c \Delta v_s$

$$\sigma_s = Max[stress(z,t)] \tag{2}$$

(1)

where ρ_0 is the initial sample density, and *c* denotes the sound speed in the sample. In fact, there exist three tensile stress peaks at different positions in the nanorod, i. e. vicinity of both end surfaces and the middle of the target region, respectively. The computational tensile strength results for different impact velocities U_f are presented in Fig.6. As is seen, the tensile stress from the right end surface σ_{s2} shows a large deviation between the examined impact velocity between 4 km/s and ¹⁷² 7 km/s. This is mainly due to the inappropriateness of acoustic approximation under strong impact, ¹⁷³ which has been discussed in previous articles [36,37]. The tensile stresses in other regions, includ-¹⁷⁴ ing the vicinity of S_F (σ_{s3}), middle of the target region (σ_{s4}), and the vicinity of S_T (σ_{s5}), are in ¹⁷⁵ the range of 3 GP and 6 GPa, and they follow a general relationship as $\sigma_{s5} > \sigma_{s4} > \sigma_{s3}$. The flyer ¹⁷⁶ rapidly transforms from solid into liquid state after impact, thus the tensile strength shown in Fig.6 ¹⁷⁷ is the minimum. The peak pressure near S_T is less than 10 GPa (Fig. 3), and σ_{s5} has the maximum ¹⁷⁸ value as expected.



Figure 6: Tensile strength of sample from bulk and end surface for different impact velocities. σ_{s1} and σ_{s2} are calculated from S_F and S_T respectively, σ_{s3} , σ_{s4} and σ_{s5} are taken from the vicinity of S_F , middle of target and the vicinity of S_T , respectively.

179 Void nucleation, evolution and closure

180 We further explore the detailed damage evolution inside the nanorod. As discussed above, the ma-

terial will form voids due to the interaction between the radial and axial release waves. Here, we

- the choose $U_f = 5$ km/s and 7 km/s cases to demonstrate the void nucleation and evolution process, as
- illustrated in Fig.7 (a) and Fig.7 (b), respectively. The void evolution can be readily divided into
- ¹⁸⁴ four stages, i.e., nucleation, growth, coalescence and closure. At the early time, the nucleation of

voids is observed and these small voids mainly appear along [110] and [110] orientations, due to 185 the strong surface effects and crystal orientations effects. Compared with $U_f = 5$ km/s, more voids 186 are generated and they appear earlier under the higher impact velocity of $U_f = 7$ km/s. The size and 187 the number of the voids increase gradually until 4.5 ps. Thereafter, voids coalesce begins, which 188 leads to the formation of bigger voids. More importantly, these voids do not develop into entire 189 spall fracture eventually like macroscopic impact. We find that the rapid expansion of impact re-190 gion will limit the growth of voids and even absorb them. After the complete disappearance of 191 voids in the target region, the voids in the flyer region at the same time transform into a very large 192 void. Such big void is unable to support the extrusion at the melting state and gradually shrink and 193 close completely at 13.0 ps.



Figure 7: Microscopic views of void evolution for: (a) $U_f = 5$ km/s; and (b) 7 km/s. For (b), the microscopic views at the time of 4.4, 6.3 and 9.3 ps correspond to the points A, B and C in Fig.5, respectively.

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To further understand the void evolution, Fig.8 compares the surface areas of total voids(S_{voids}) be-195 tween $U_f = 5$ km/s and 7 km/s. As is seen, the surface area starts to increase at 2.5 ps and 1.9 ps 196 for $U_f = 5$ km/s and 7 km/s, respectively, indicating the nucleation of voids. With continuing sim-197 ulation, the surface area increases significantly, and reaches the maximum at 4.5 ps and 4.4 ps for 198 U_f = 5 km/s and 7 km/s, respectively. Specifically, the maximum voids surface area under U_f =7 199 km/s is almost two times of that under $U_f = 5$ km/s. After that, it decreases remarkably due to the 200 voids coalescence. It is noted that there is a clear inflection on the surface area profile during the 201 decreasing period, which is caused by the formation of bigger voids. The surface area decreases to 202

²⁰³ zero eventually, due to the closure of voids. Compared with $U_f = 5$ km/s, the voids appear earlier ²⁰⁴ and last for longer under the higher impact velocity of $U_f = 7$ km/s.



Figure 8: Variation of the surface areas of total voids with time for the cases of $U_f = 5$ km/s and 7 km/s, respectively. A, B, C, D represent the nucleation, maximal surface area, coalescence, and complete closing of voids, respectively. An obvious inflection on the curves can be seen, implying that the convergent holes begin to close.

Radial expansion and fragmentation

The evolution of radial average velocity ($v_r = \sqrt{v_x^2 + v_y^2}$) of impact plane is presented in Fig.9, 206 which provide important information at the early stage of radial expansion. Here, a cylindrical sec-207 tion on the impact plane with a thickness of 10a is selected to analyze the impact process for the 208 cases of $U_f = 4$ km/s and 7 km/s, respectively. To note that atoms at the edge of the impact layer 209 first gain velocity rather than atoms in the center region due to the strong side surface release ef-210 fect. As shown in Fig.9 (a) and Fig.9 (b), the radial velocity at the time of 0.5 ps is almost a con-211 stant from the center to a radial distance of around 10 nm, and it increases sharply nearly the edge. 212 With further simulation, the radial velocity of the central area increase, and results in a gradual in-213 creasing profile. Interestingly, the radial velocity profile shows a distinct discontinuity point at the 214

position of the initial radius (highlighted by the dashed magenta lines in Fig.9 (a) and Fig.9 (b)).

As expected, the impact layer expands faster under $U_f = 7$ km/s and the velocity of the edge atoms is higher than the atoms at center.



Figure 9: The radial velocity distributions in the impact plane for (a) $U_f = 4$ km/s and (b) 7 km/s at different moments. The microscopic views demonstrate the fragmentation process and for better visualization those views have been scaled differently. The results for $U_f = 4$ km/s show an evident orientation dependence and the results for $U_f = 7$ km/s seem to be a ring breakup. And the former has very limited and larger clusters, than the latter. The atoms are colored by their average speed *v*.

The impact plane continues to expand and forms many fragments. To analysis the structural evolution, we decompose the sample into disconnected clusters (that contains a number of atoms) based on a cutoff radius that equals to the lattice constant. The typical pictures of cluster evolution, in-

cluding formation, expansion and fragmentation under the conditions of $U_f = 4$ km/s and 7 km/s 221 are shown in Fig.9 (c) and Fig.9 (d). According to Fig.9 (c) under the velocity of U_f =4 km/s, the 222 front part of the target region remains in a solid state during the impact process, while the other 223 regions are like in a melt state. Due to the rapid outward expansion, fragmentation appears at the 224 vicinity of the edges of the impact layer (due to the different expansion rates). The debris spread 225 from the center to the periphery and exhibit distinct shapes (at the time of 100 ps in Fig.9 (c)). It 226 is interesting to note that the formed fragments or debris are along the [110] and [010] crystalline 227 directions. Under high impact velocity ($U_f = 7$ km/s, Fig.9 (d)), similar fragmentation process is 228 observed. The difference is much earlier separation between the central area and the edges is ob-229 served (at 20 ps). Due to the high kinetic energy, the sample eventually decomposes into a larger 230 number of small clusters. 231

We further estimate the number and size of clusters using the approach as utilized in literature[38, 39]. The distribution of number and size of clusters under different impact velocities U_f exhibits similar features. As shown in Fig.10, the size distribution manifests a remarkable power laws of follow form:

$$N_c(N_p) \propto N_p^{\ k} \tag{3}$$

where N_p is the number of atoms in each cluster and N_c is the number of clusters corresponding to 237 N_p . The solid line in Fig.10 represents the power law between N_c and N_p , which is fitted from the 238 final state of different impact processes. We use the green dash line for the case of $U_f = 4$ km/s, 239 since the number of clusters here is too small. |k| seems to increase with the increase of U_f , which 240 increases from 1.49 to 1.87 when U_f increases from 5 km/s to 7 km/s. Notice that $N_p < 1000$ is 241 neglected for the fitting, which is considered as insignificant in the fragmentation process. The dis-242 tribution has similarity with ejecta size distribution[39] and the results can reveal some regulations 243 of clusters evolution during hypervelocity impact at nanoscale. 244



Figure 10: Relationship between the number of atoms in each cluster N_p and the corresponding number of clusters N_c at different impact velocities U_f .

245 Summary

In this work, we conduct MD simulations to investigate the damage characteristic of single crys-246 tal aluminum nanorod under hypervelocity impact. The simulation results demonstrate some in-247 teresting dynamic properties. During the propagation process, the impact pressure experiences a 248 rapid decaying due to its release from the side surface, which is reflected by the pressure and ve-249 locity profiles along the axial direction. The interaction of impact wave and its release wave from 250 side and end surfaces result in a complex three-dimensional stress state. Two tension regions in the 251 nanorod are formed, which lead to the nucleation of voids. The evolution of voids can lead to the 252 oscillation of flyer free surface velocity to some extent and the tensile strength of nanorod is sim-253 ilar with that of bulk aluminum from our simulations. However, all these voids are temporary and 254 they will close after they grow and coalesce, therefore no complete spall fracture occurs ultimately. 255 Such observation is due to the increasing new surface release effect limits the expansion of voids 256 and the velocity gradient can easily facilitate the voids to collapse at melting state. Besides, the 257 void nucleation is preferably along [110] and [$1\overline{1}0$]. Additionally, it is found that the cluster for-258

mation under low impact velocity is found to exhibit an apparent orientation dependence related to
the early damage evolution, and at final state each cluster is close to a sphere due to the surface tension. Specifically, the cluster number is found to follow a power law relationship with the cluster
size.

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